

### A Mixing Model for the Richtmyer-Meshkov Instability

A. W. Cook, C. Weber, W. H. Cabot, R. Bonazza

June 8, 2012

IWPCTM13 Conference Wolburn, United Kingdom July 16, 2012 through July 20, 2012

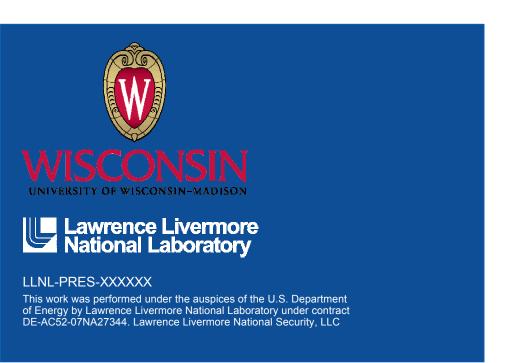
#### Disclaimer

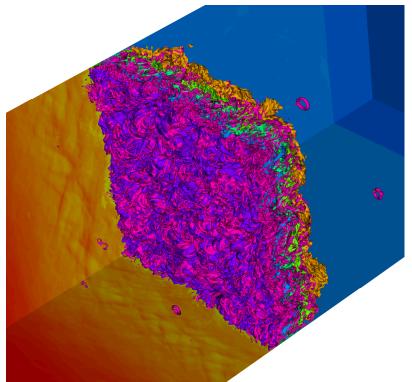
This document was prepared as an account of work sponsored by an agency of the United States government. Neither the United States government nor Lawrence Livermore National Security, LLC, nor any of their employees makes any warranty, expressed or implied, or assumes any legal liability or responsibility for the accuracy, completeness, or usefulness of any information, apparatus, product, or process disclosed, or represents that its use would not infringe privately owned rights. Reference herein to any specific commercial product, process, or service by trade name, trademark, manufacturer, or otherwise does not necessarily constitute or imply its endorsement, recommendation, or favoring by the United States government or Lawrence Livermore National Security, LLC. The views and opinions of authors expressed herein do not necessarily state or reflect those of the United States government or Lawrence Livermore National Security, LLC, and shall not be used for advertising or product endorsement purposes.

### A Mixing Model for the Richtmyer-Meshkov Instability

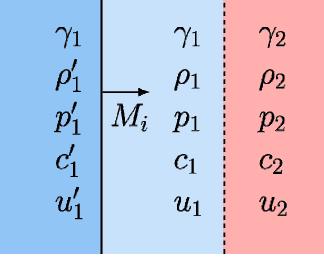
IWPCTM 16-20 July 2012

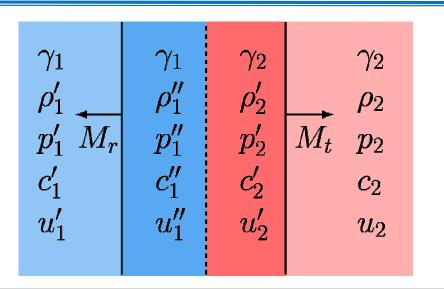
Andy Cook, Chris Weber, Bill Cabot & Riccardo Bonazza

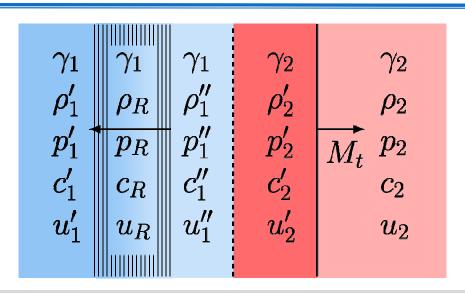




## Consider a shock crossing an interface between two ideal gases







# Matching pressures and velocities across the interface yields a transcendental equation for the pressure jump across the transmitted shock

$$\left[\frac{(\Lambda_2 - 1)\rho_1}{(\Lambda_1 - 1)\rho_2}\right]^{1/2} \frac{\Pi_t - 1}{(\Pi_t \Lambda_2 + 1)^{1/2}} = \frac{\Pi_i - 1}{(\Pi_i \Lambda_1 + 1)^{1/2}} - \left(\frac{\rho_1}{\rho_1'}\right)^{1/2} \frac{\Pi_t - \Pi_i}{(\Pi_t \Lambda_1 + \Pi_i)^{1/2}}$$

$$\Pi_i \equiv \frac{p_1'}{p_1} \,, \quad \Lambda_1 \equiv \frac{\gamma_1 + 1}{\gamma_1 - 1}$$

$$\Pi_t \equiv \frac{p_2'}{p_2} \;, \quad \Lambda_2 \equiv \frac{\gamma_2 + 1}{\gamma_2 - 1}$$

All other variables are readily obtained after solving this equation.

## Does the RM growth rate obey a power law at late time?

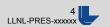
$$h = ct^{\theta}$$

$$h \propto (u_s t)^{\theta}$$

$$h - h_o \propto (t - t_o)^{\theta}$$

$$0.2 \le \theta \le 0.67$$

- What are the dimensions of c?
- How is this growth rate derived?
- Is it a good idea to raise dimensional variables to fractional powers?
- What's missing?



## The mixing with (h) and time (t) must be properly nondimensionalized

- We can eliminate the virtual origin by modeling h, rather than h.
- Normalizing  $\dot{h}$  by its initial value,  $h_o$ , ensures that all growth curves start at unity.
- Linear stability theory and experimental evidence indicate that the growth rate depends on the dominant perturbation wavelength  $\lambda_{\alpha}$ .
- A relevant timescale thus appears to be  $\lambda_o$  /  $h_o$ .

### How do we get $h_o$ ?

- Assume interfacial perturbations are known.
- Define  $h(t)\equiv\int_{-\infty}^{\infty}\psi(\langle\xi\rangle)\;dx$  , where  $\psi$  is "product". h(t) is the thickness of mixed fluid that would result
- h(t) is the thickness of mixed fluid that would result if the entrained gases were homogenized in the transverse plane.
- From continuity, growth rate = mass flux through equimolar plane:

$$\frac{dh}{dt} = 2 \int_{-\infty}^{x_s} \frac{\partial \langle \xi \rangle}{\partial t} dx - 2 \int_{x_s}^{\infty} \frac{\partial \langle \xi \rangle}{\partial t} dx = \frac{4 \langle \rho u \rangle|_{x_s}}{\rho_1'' - \rho_2'}$$

## The mass-flux definition of the growth rate has significant advantages:

- No issues of asymmetry between bubbles and spikes.
- Not sensitive to outliers (like threshold definitions).
- Valid for shocks in either direction.
- Data need only be gathered on a single plane (PLIF friendly).

## The post-shock density field is obtained directly from the known perturbations

$$\dot{h}_o \approx \dot{h}^+ \equiv \frac{4\langle \rho^+ u^+ \rangle_{x^+}}{\rho_1^{"} - \rho_2^{'}}$$

$$\rho^{-}(x, y, z) = \rho_1 + (\rho_2 - \rho_1) H(x - \eta(y, z))$$

$$\rho^{+}(x,y,z) = \rho_{1}'' + (\rho_{2}' - \rho_{1}'') H(x - \eta^{+}(y,z))$$

$$\eta^+ = \left(1 - \frac{u_s}{M_i c_1}\right) \eta + x^+$$

# The post-shock velocity field can be obtained from Biot-Savart integration of the vorticity field

$$\mathbf{u}^{+}(\mathbf{x}) = \frac{1}{4\pi} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \frac{\boldsymbol{\omega}^{+}(\mathbf{x}^{*}) \times (\mathbf{x} - \mathbf{x}^{*})}{||\mathbf{x} - \mathbf{x}^{*}||^{3}} dx^{*} dy^{*} dz^{*}$$

$$\frac{\partial \boldsymbol{\omega}}{\partial t} + \boldsymbol{\nabla} \times (\boldsymbol{\omega} \times \boldsymbol{u}) = \frac{1}{\rho^2} \boldsymbol{\nabla} \rho \times \boldsymbol{\nabla} p + \boldsymbol{\nabla} \times \left( \frac{1}{\rho} \boldsymbol{\nabla} \cdot \underline{\boldsymbol{\tau}} \right)$$

$$\frac{\partial \boldsymbol{\omega}}{\partial t} \approx \frac{1}{\rho^2} \boldsymbol{\nabla} \rho \times \boldsymbol{\nabla} p$$

## The pressure gradient can be obtained by assuming an essentially planar shock

$$\frac{\partial p}{\partial x} \approx -\frac{\partial \rho u}{\partial t}$$

$$\frac{\partial p}{\partial y} \approx 0$$

$$\frac{\partial p}{\partial z} \approx 0$$

$$\frac{\partial \omega_x}{\partial t} \approx 0$$

$$\frac{\partial \omega_y}{\partial t} \approx -\frac{1}{\rho^2} \frac{\partial \rho}{\partial z} \frac{\partial \rho u}{\partial t}$$

$$\frac{\partial \omega_z}{\partial t} \approx \frac{1}{\rho^2} \frac{\partial \rho}{\partial y} \frac{\partial \rho u}{\partial t}$$

### The impulsive approximation allows us to separate the spatial and temporal dependence of density and velocity

$$\rho \approx \rho^{-} + (\rho^{+} - \rho^{-})\mathcal{H}(t)$$
$$u \approx u_{s}\mathcal{H}(t)$$

$$\rho \approx \rho^{-} + (\rho^{+} - \rho^{-})\mathcal{H}(t)$$

$$u \approx u_{s}\mathcal{H}(t)$$

$$\frac{\partial \rho}{\partial z} = \frac{\partial \rho^{-}}{\partial z} + \left(\frac{\partial \rho^{+}}{\partial z} - \frac{\partial \rho^{-}}{\partial z}\right)\mathcal{H}(t)$$

$$\frac{\partial \rho}{\partial y} = \frac{\partial \rho^{-}}{\partial y} + \left(\frac{\partial \rho^{+}}{\partial y} - \frac{\partial \rho^{-}}{\partial y}\right)\mathcal{H}(t)$$

$$\frac{\partial \rho u}{\partial t} = \rho^{+}u_{s}\delta(t) .$$

## The vorticity laid down by the shock is obtained by integrating over the impulse

$$\omega_x^+ \equiv \int_{0^-}^{0^+} \frac{\partial \omega_x}{\partial t} \, dt \approx 0$$

$$\omega_y^+ \equiv \int_{0^-}^{0^+} \frac{\partial \omega_y}{\partial t} dt \approx -\frac{u_s}{\rho^+} \frac{\partial \rho^+}{\partial z}$$

$$\omega_z^+ \equiv \int_{0^-}^{0^+} \frac{\partial \omega_z}{\partial t} dt \approx \frac{u_s}{\rho^+} \frac{\partial \rho^+}{\partial y}$$

## The post-shock vorticity can be written directly in terms of the perturbations

$$\omega_y^+ \approx \frac{u_s (\rho_2' - \rho_1'') \frac{\partial \eta^+}{\partial z} \delta(x - \eta^+)}{\rho_1'' + (\rho_2' - \rho_1'') H(x - \eta^+)}$$

$$\omega_z^+ \approx -\frac{u_s (\rho_2' - \rho_1'') \frac{\partial \eta^+}{\partial y} \delta(x - \eta^+)}{\rho_1'' + (\rho_2' - \rho_1'') H(x - \eta^+)}$$

## The post-shock velocity can thus be computed a priori

$$u^{+} \approx \frac{u_{s}A^{+}}{2\pi} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \frac{\frac{\partial \eta^{+}(y^{*},z^{*})}{\partial y^{*}}(y-y^{*}) + \frac{\partial \eta^{+}(y^{*},z^{*})}{\partial z^{*}}(z-z^{*})}{\left[(x-\eta^{+}(y^{*},z^{*}))^{2} + (y-y^{*})^{2} + (z-z^{*})^{2}\right]^{3/2}} dy^{*} dz^{*}$$

How good is the impulsive-planar-shock approximation?

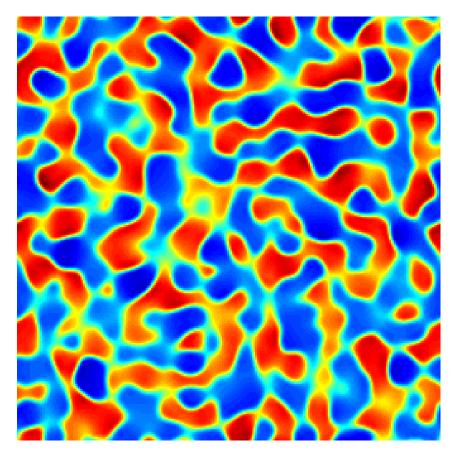


## Mass flux ( $\rho u$ ) on the equimolar plane, $\langle \xi \rangle = 0.5$

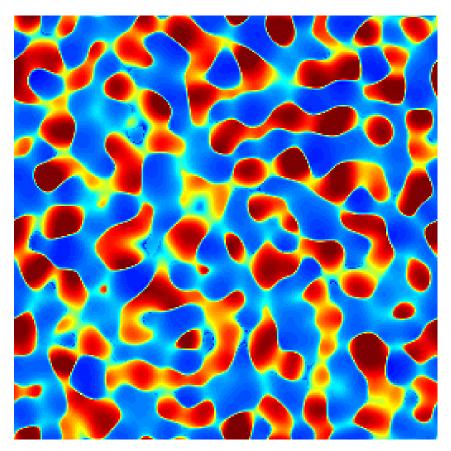
$$A = 0.53$$

$$M_i = 1.1$$

$$\eta_{rms} / \lambda_o = 0.1$$

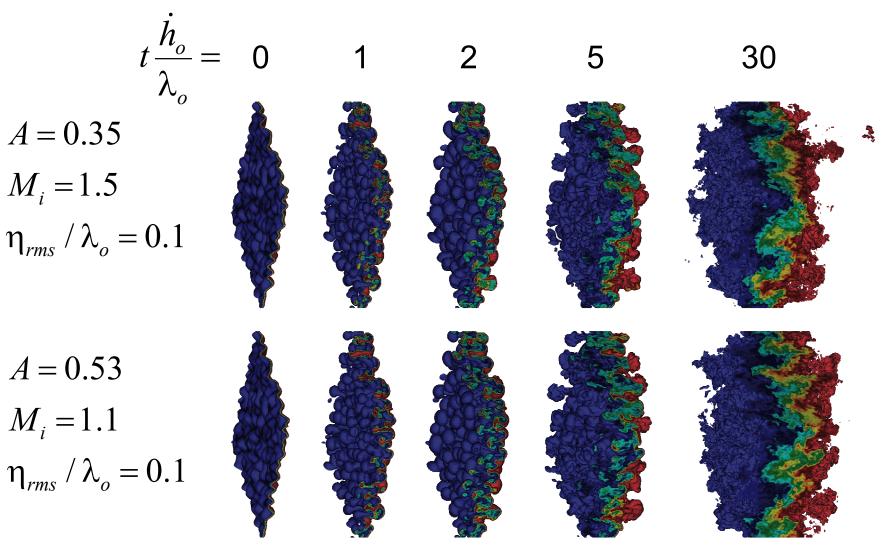


Simulation

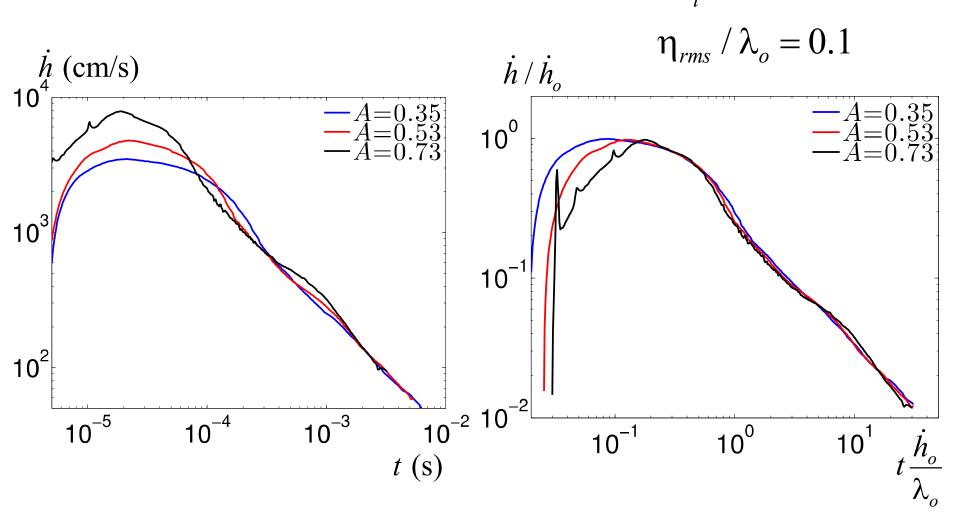


Model

### Visualization at nondimensional times



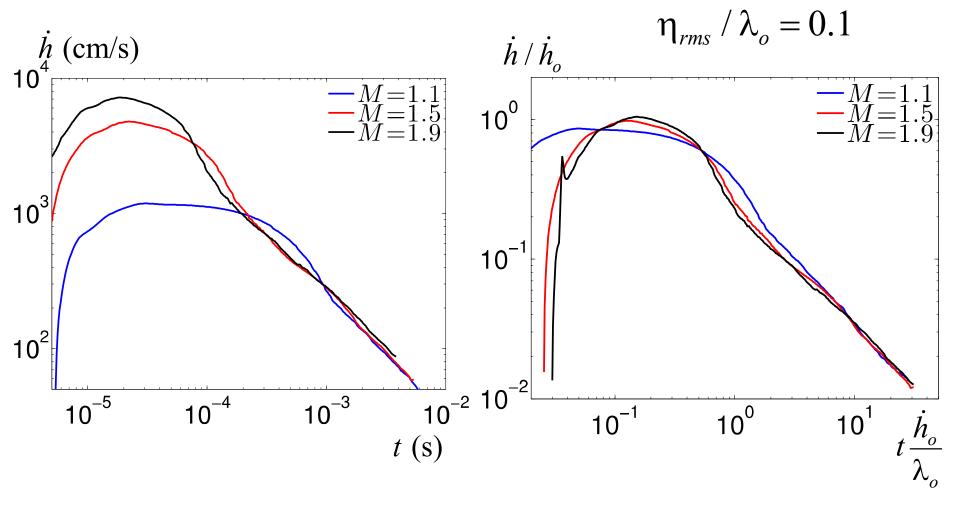
## The scaled growth rates collapse for different Atwood numbers (A > 0) $M_i = 1.5$



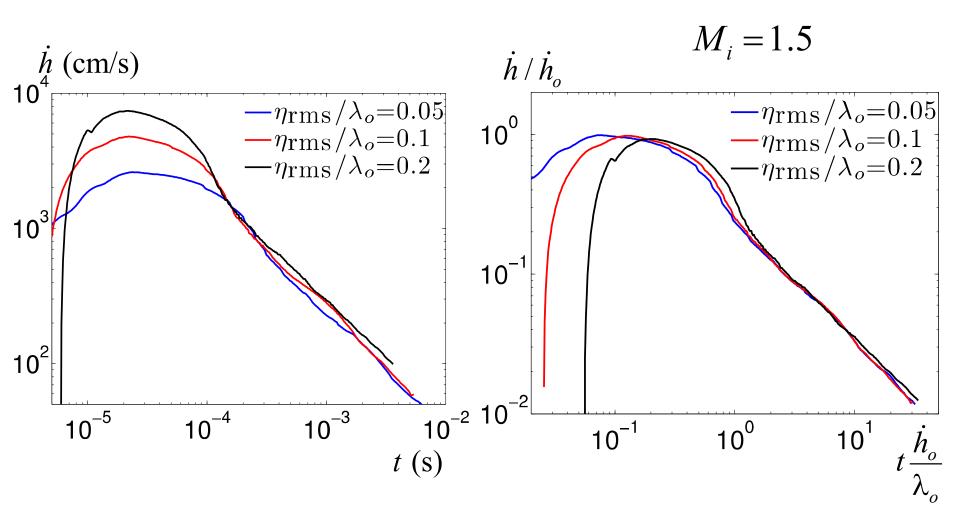
### The scaled growth rates collapse for different

**Mach numbers** 

$$A = 0.53$$



## The scaled growth rates collapse for different amplitude/wavelength ratios A = 0.53



### Time axis shift for heavy-to-light cases (A < 0)

Interface thickness:

(note that 
$$\dot{h}_o < 0$$
 for  $A < 0$ )

$$h = \dot{h}_o t + h_o$$

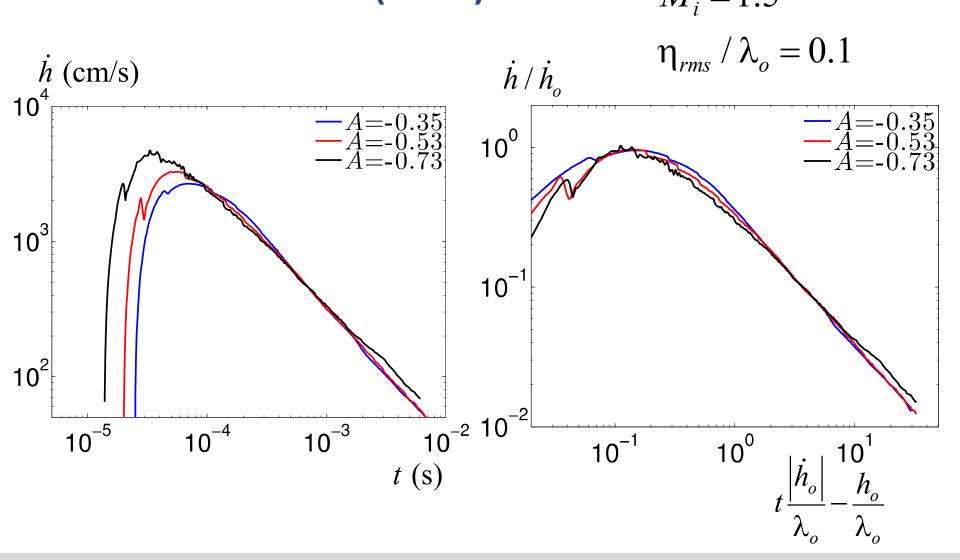
Thickness is minimum at:

$$h \approx 0$$
  $t = \frac{-h_o}{\dot{h}_o}$ 

Shift time axis by this amount (phase inversion time)

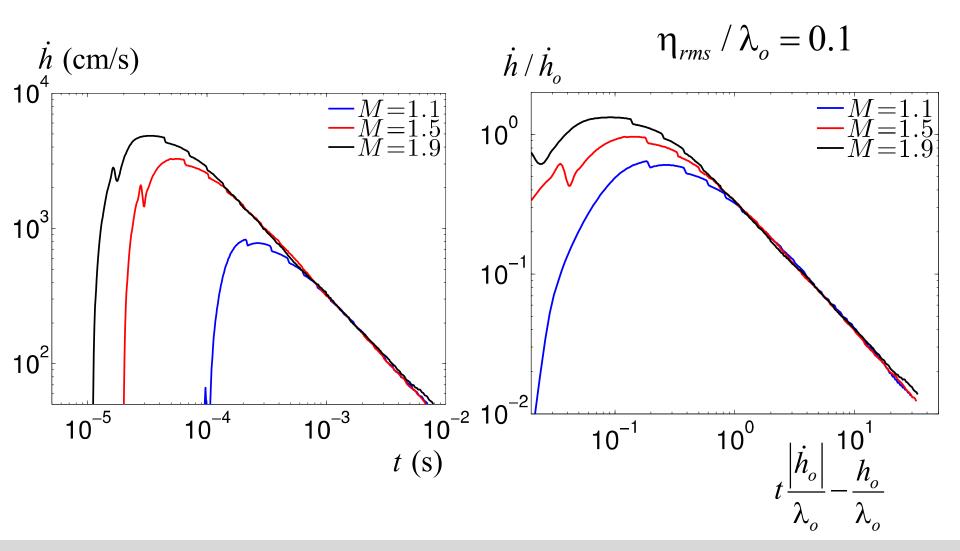
$$t - \frac{-h_o}{\dot{h}_o}$$
 nondimensional  $t \frac{|\dot{h}_o|}{\lambda_o} - \frac{h_o}{\lambda_o}$ 

## The scaled growth rates collapse for different Atwood numbers (A < 0) $M_i = 1.5$

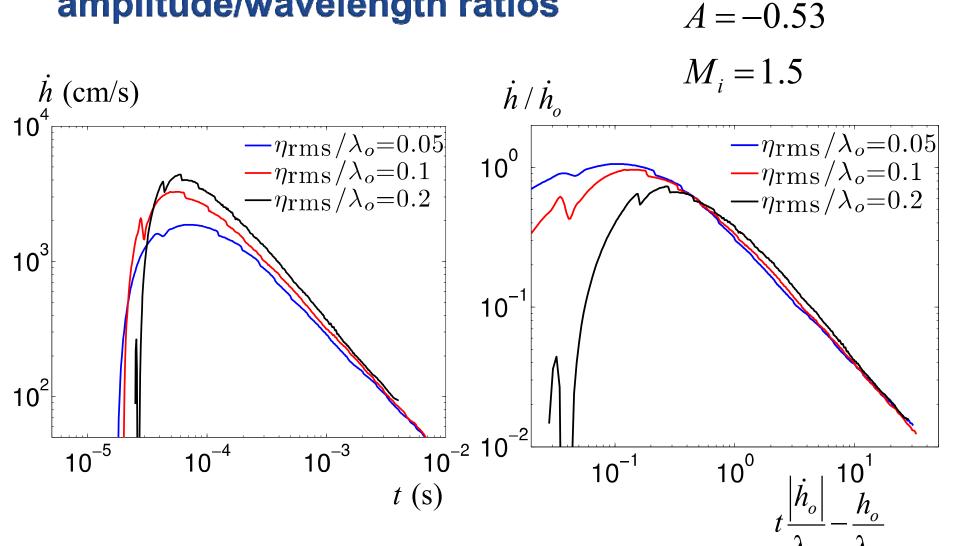


### The scaled growth rates collapse for different **Mach numbers**

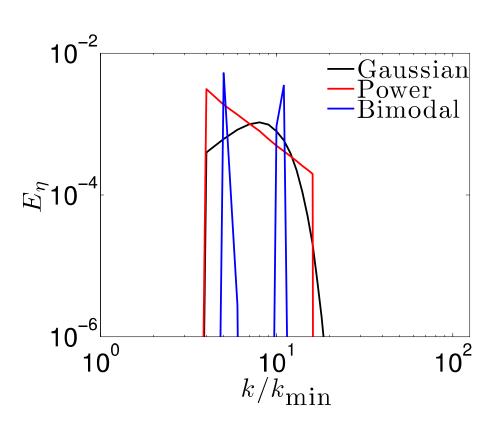
A = -0.53



### The scaled growth rates collapse for different amplitude/wavelength ratios 4 - 0.53



### Do the growth rate curves collapse for different spectral shapes?



#### Gaussian

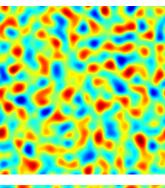
$$k_{\text{peak}} / k_{\text{min}} = 32$$

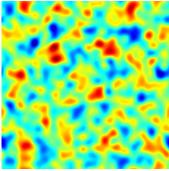
#### **Power Law**

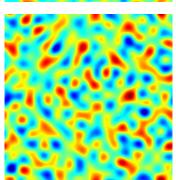
$$E_{\eta} \propto k^{-2}$$



$$k_{\text{peak}} / k_{\text{min}} = 24 \& 48$$







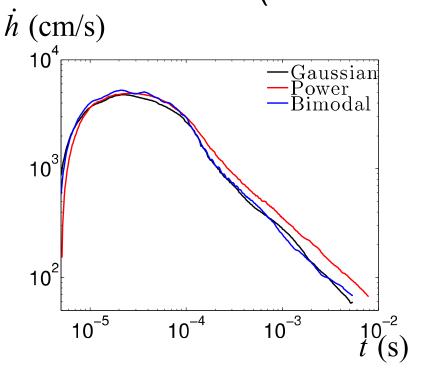
### The scaled growth rates collapse for different perturbation spectra

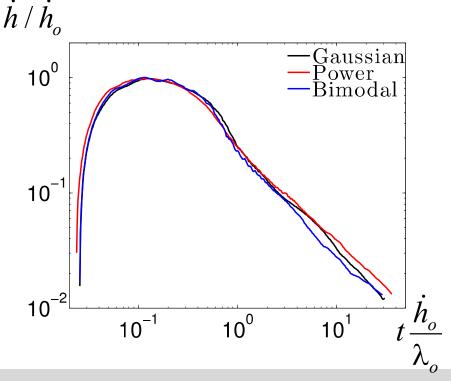
- Possible different behavior at late times
- Initial perturbation spectra widths are all rather narrow (less than a decade)

$$A = 0.53$$

$$M_i = 1.5$$

$$\eta_{rms} / \lambda_o = 0.1$$





### Collapse of the growth rate curves suggests the thickness/growth rate history can be represented by a single equation

Growth rate curves at later times fit the form

$$\frac{\dot{h}}{\dot{h}_{o}} = c\theta \tau^{\theta - 1} \xrightarrow{\text{integrating}} \frac{h - h_{o}}{\lambda_{o}} = c\tau^{\theta}$$

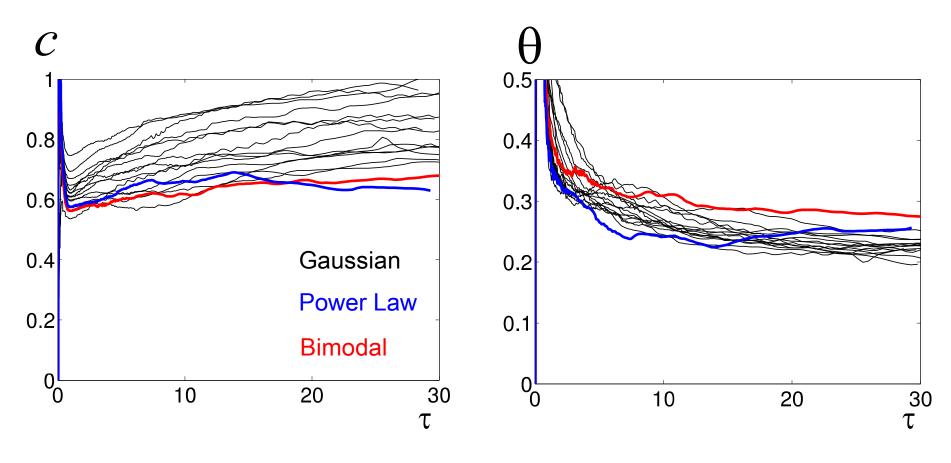
$$\frac{\dot{h}}{\lambda_{o}} = c\tau^{\theta}$$

$$t\frac{\dot{h}_{o}}{\lambda_{o}} \qquad \text{for } A > 0$$
Solve for the unknowns
$$t\frac{\dot{h}_{o}}{\lambda_{o}} - \frac{h_{o}}{\lambda_{o}} \qquad \text{for } A < 0$$

Solve for the unknowns

$$\theta = \frac{\dot{h}}{\left|\dot{h}_{o}\right|} \frac{\lambda_{o}}{h - h_{o}} \tau \qquad c = \frac{h - h_{o}}{\lambda_{o}} \tau^{\left(-\frac{\dot{h}}{\left|\dot{h}_{o}\right|} \frac{\lambda_{o}}{h - h_{o}} \tau\right)}$$

#### **Growth rate coefficients**

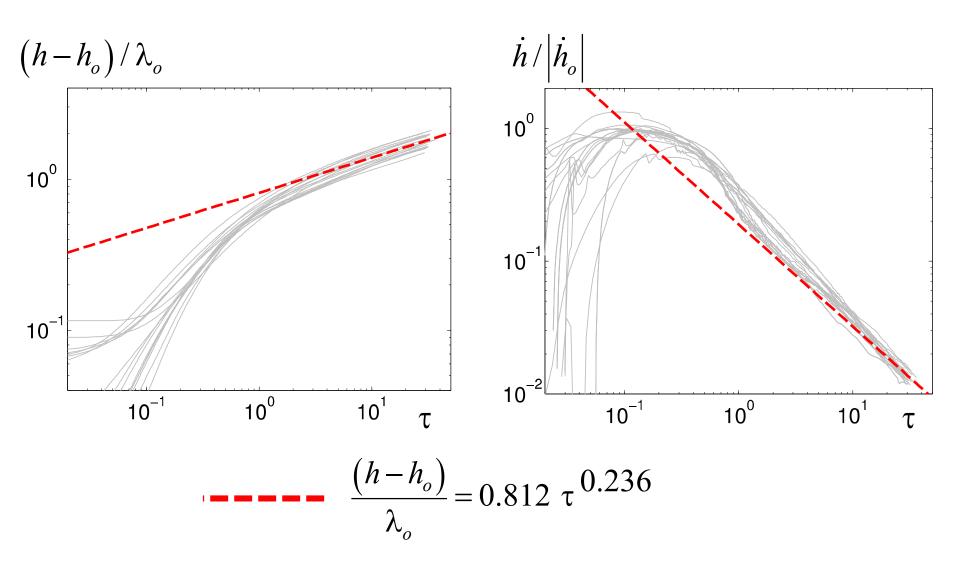


Average beyond  $\tau > 20$  c = 0.813

$$c = 0.813$$

$$\theta = 0.233$$

#### Curve fit matches the later time data well



### **Conclusions**

- The growth rate of the mixing region is determined solely by the net mass flux through the equimolar plane.
- The post-shock density and velocity fields (and hence the initial mass flux) can be accurately modeled if the interfacial perturbations are known.
- The initial growth rate (computed a priori) can be used to collapse the mixing curves for various Atwood numbers, Mach numbers etc.
- The collapse of the growth curves (and hence the universality of the scaling) may depend on whether the initial spectrum is narrow or broadband.
- A universal value of  $\theta$  may only exist for perturbation spectra of the same form.

### **Extra Slides**

### **Simulation Setup**

Gaussian Perturbation Spectrum

$$E_{\eta}(k) \propto \exp\left(\frac{-(k-k_p)^2}{k_b^2}\right)$$

Perturbation energy

$$\eta_{\rm RMS}^2 = \langle \eta^2 \rangle = \int_0^\infty E_{\eta}(k) \, dk$$

Dominant wavelength

$$\lambda_0 \equiv 2\pi \frac{\int_0^\infty E(k)/k \ dk}{\int_0^\infty E(k) \ dk} \ .$$

Interface profile

$$\xi(x, y, z) = \frac{1}{2} \left( 1 + \operatorname{Erf}\left(\frac{d(x, y, z)}{\sigma}\right) \right)$$

Distance function

$$d(x, y, z) = sign(x - \eta(\bar{y}, \bar{z})) \min_{\bar{y}, \bar{z}} \left( \sqrt{(x - \eta(\bar{y}, \bar{z}))^2 + (y - \bar{y})^2 + (z - \bar{z})^2} \right)$$

### **Simulation Resolution**

- By  $256^2 \times 512$ , the peak growth rate is within 2% of the modeled  $h_o$
- A k<sup>-5/3</sup> inertial range develops at the two highest resolutions

A = 0.53

$$M_i = 1.5$$

 $\eta_{rms} / \lambda_o = 0.1$ 

